MQ-9 Propeller Optimization

PropOps

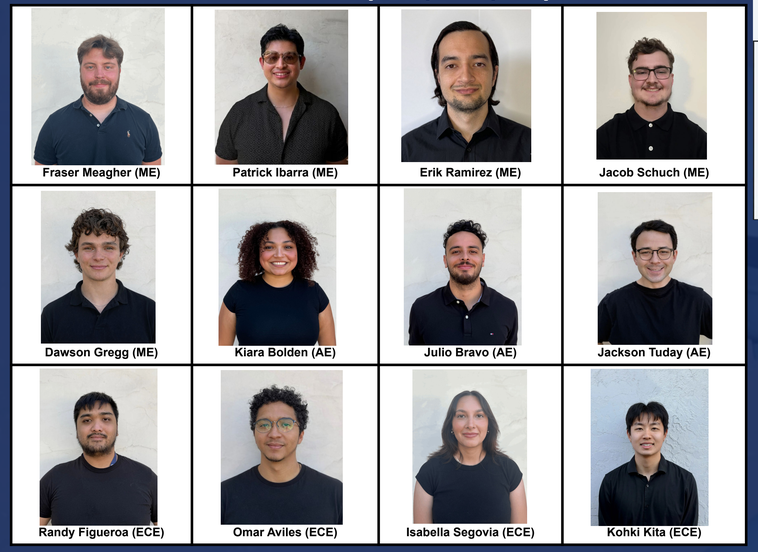
San Diego State University

GENERAL ATOMICS AERONAUTICAL

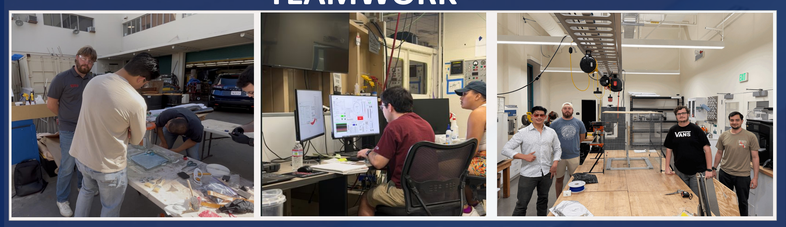
PROJECT OVERVIEW

The MQ-9 Reaper is a long-endurance unmanned aircraft where propeller efficiency and acoustic signature are critical to mission performance. This project focuses on the design, fabrication, and testing of an optimized propeller system to improve aerodynamic efficiency while reducing noise. A scaled propeller model and modular test stand were developed to experimentally evaluate performance. The system enables measurement of thrust, torque, RPM, velocity, and sound characteristics. The goal is to analyze tradeoffs between efficiency and noise reduction and provide data-driven design improvements relative to a baseline propeller. The final phase is manufacturing a propeller assembly out of aerospace grade materials.

THE TEAM (ME,ECE,AE)



TEAMWORK



ABOUT OUR SPONSORS

General Atomics Aeronautical Systems is a leader in unmanned aircraft systems like the MQ-9 Reaper

ACKNOWLEDGMENTS

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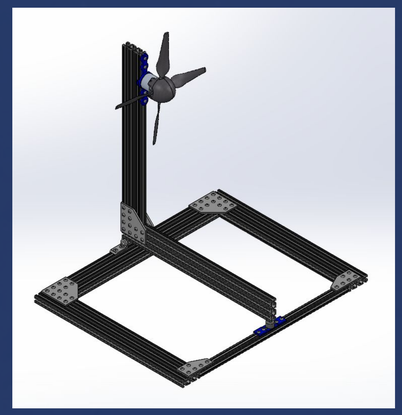
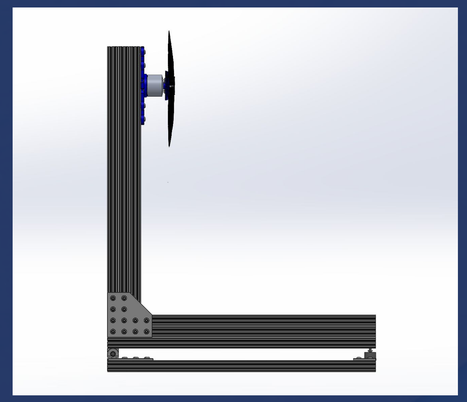
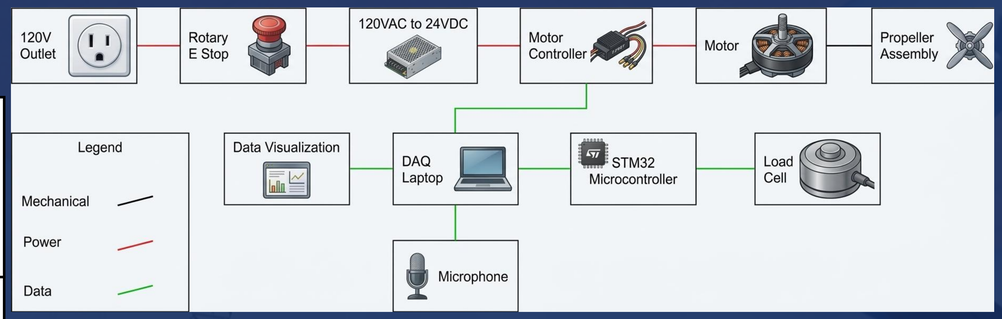
TEST STAND

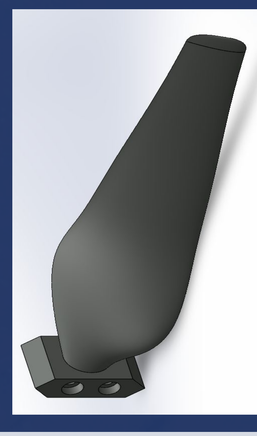




SYSTEM LEVEL DIAGRAM



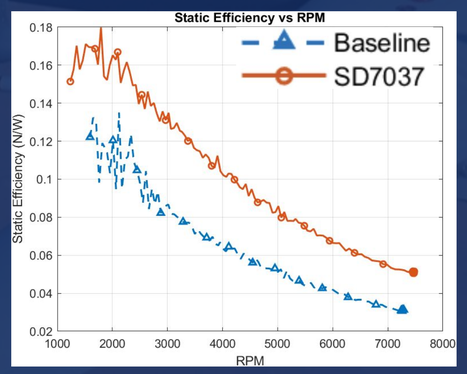
PROPELLER DESIGNS



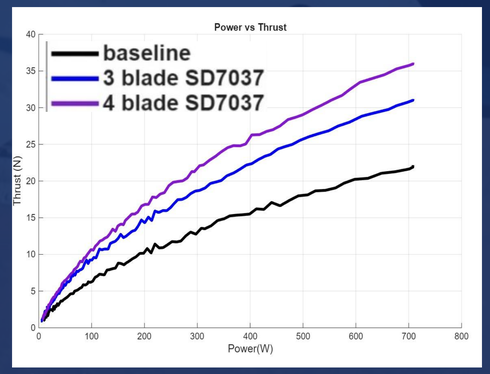
Aerospace-grade blades and hub were fabricated using forged carbon fiber, a carbon fiber spinner, and a machined aluminum hub. The **full assembly (left)**, **GA-ASI baseline (middle)**, and **SD7037 V2 (right)** are shown, with the baseline used for consistent performance comparison. The SD7037 V2 was developed through a collaborative, iterative design process, initially guided by BEMT and subsequently refined using CFD and experimental testing, resulting in improved efficiency.

RESULTS

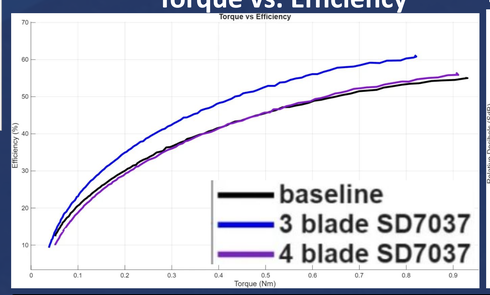
Static Efficiency vs. RPM



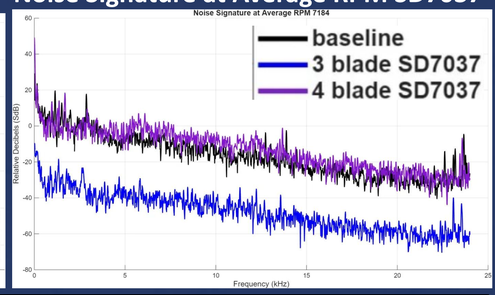
Power vs. Thrust



Torque vs. Efficiency



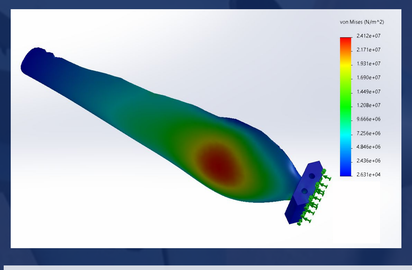
Noise Signature at Average RPM SD7037



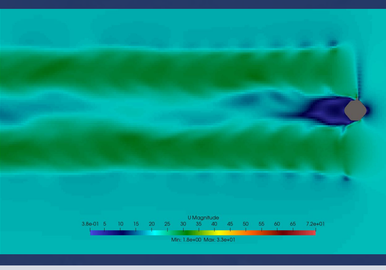
Measured noise signatures show blade count affects acoustics. The four-blade configuration produced lower peak noise and a smoother spectrum than the three-blade, indicating improved uniformity. Overall, the four-blade is preferable for reduced acoustic signature despite aerodynamic tradeoffs.

ANALYSIS METHODS

FEA



CFD



Finite Element Analysis (FEA) using von Mises stress identified peak blade maximum stresses under maximum thrust, confirming structural integrity with adequate safety margin. **Computational Fluid Dynamics (CFD)** evaluated aerodynamic performance across RPMs, capturing pressure distribution and tip vortices. Results guided blade refinement and validated BEMT predictions, confirming improved efficiency in the SD7037 V2 design.

MANUFACTURING

3D Printing Propellers (Left) and Carbon Fiber Forged Propeller (Right)

Spring 2026