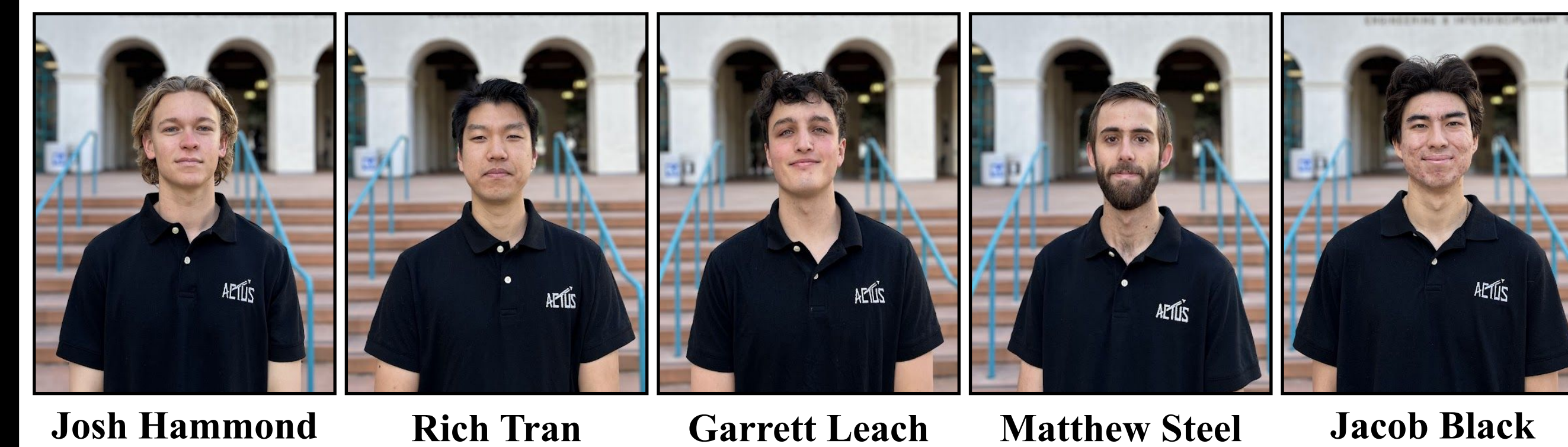


## Electric Actuation for Retractable Landing Gear System

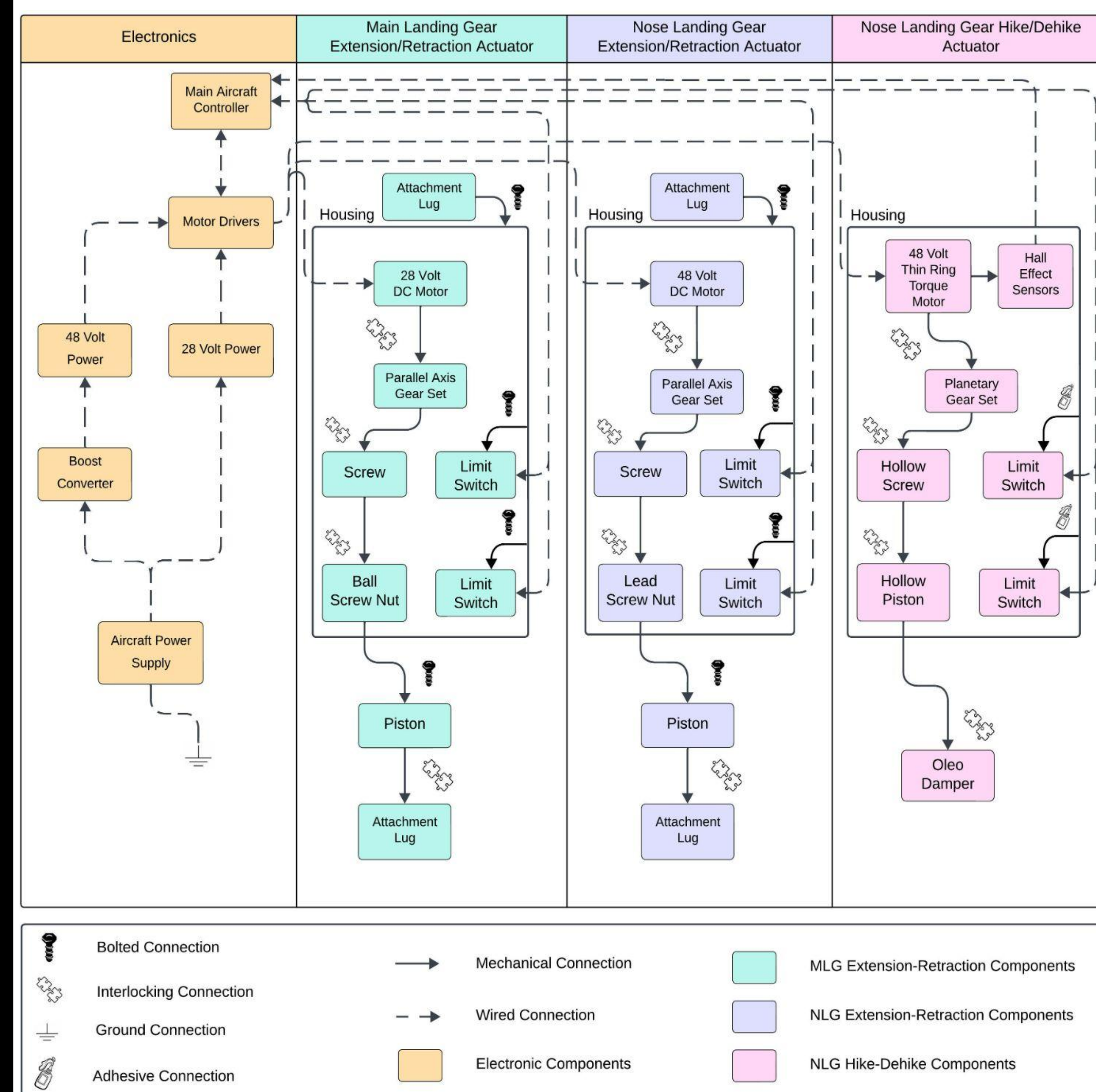
### Description

Many unmanned aircraft utilize electric power for systems that, on manned aircraft, have traditionally been hydraulically powered. This change can reduce overall aircraft weight and facilitate computer control of the system(s). For this project, the team was tasked with designing, prototyping, and testing an electric actuation system for an existing hydraulically actuated landing gear configuration.

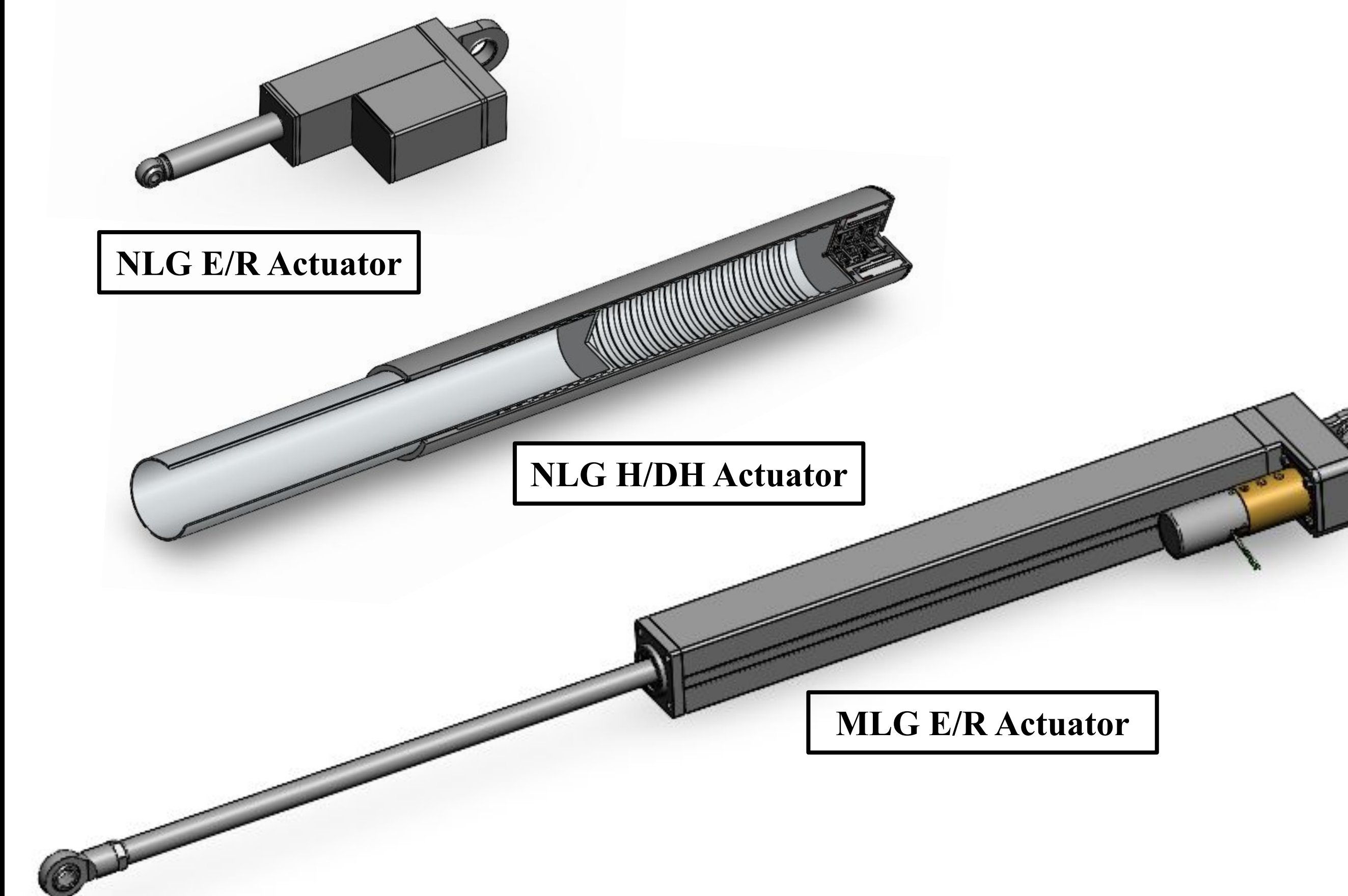
### Team Members



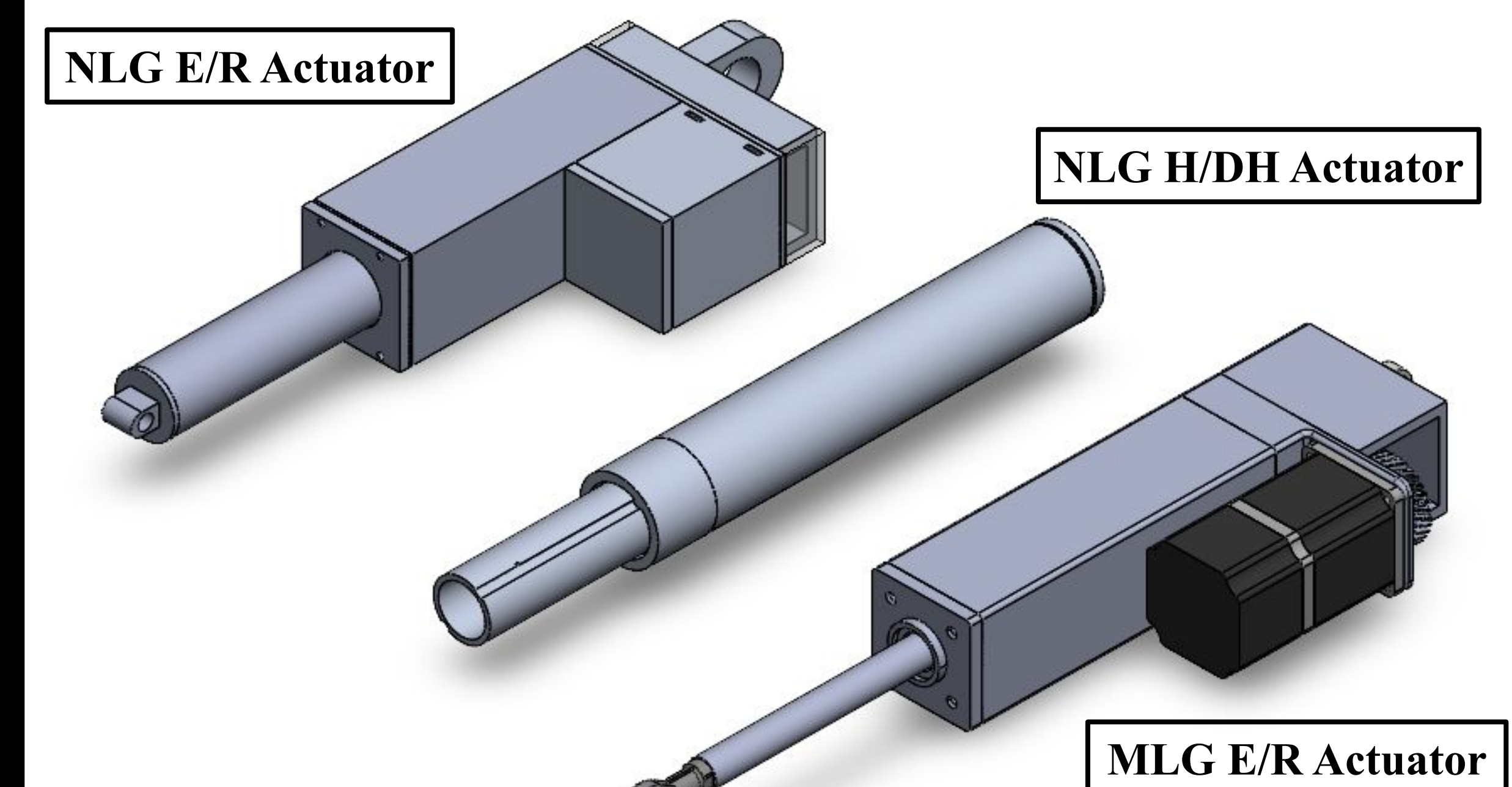
### System Level Diagram



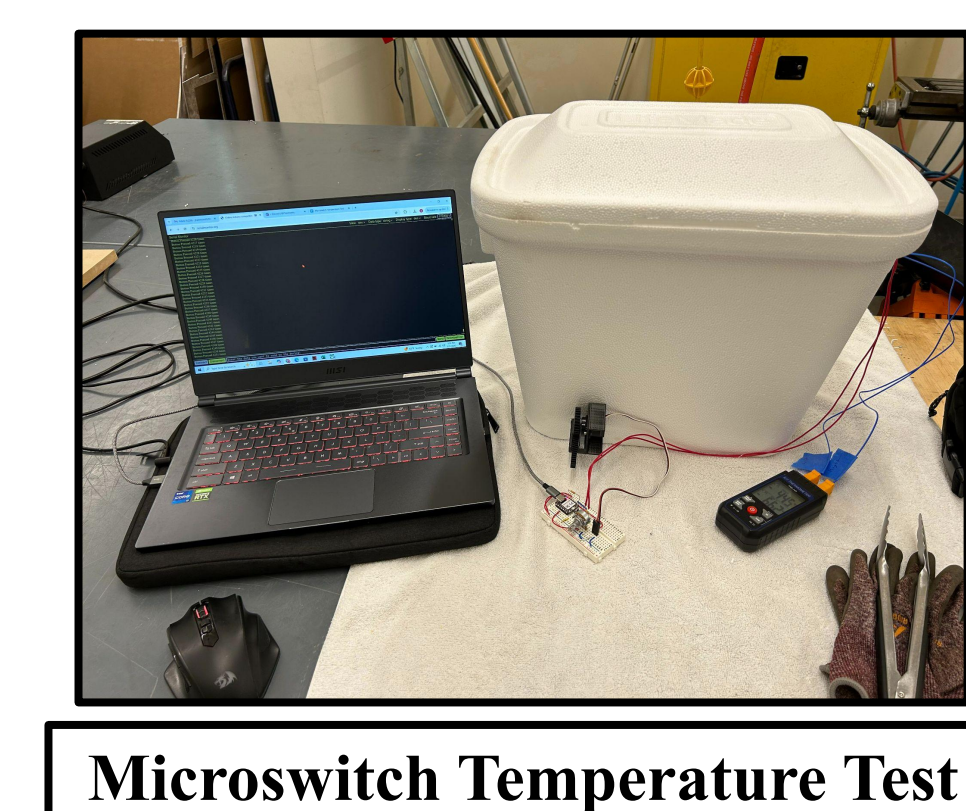
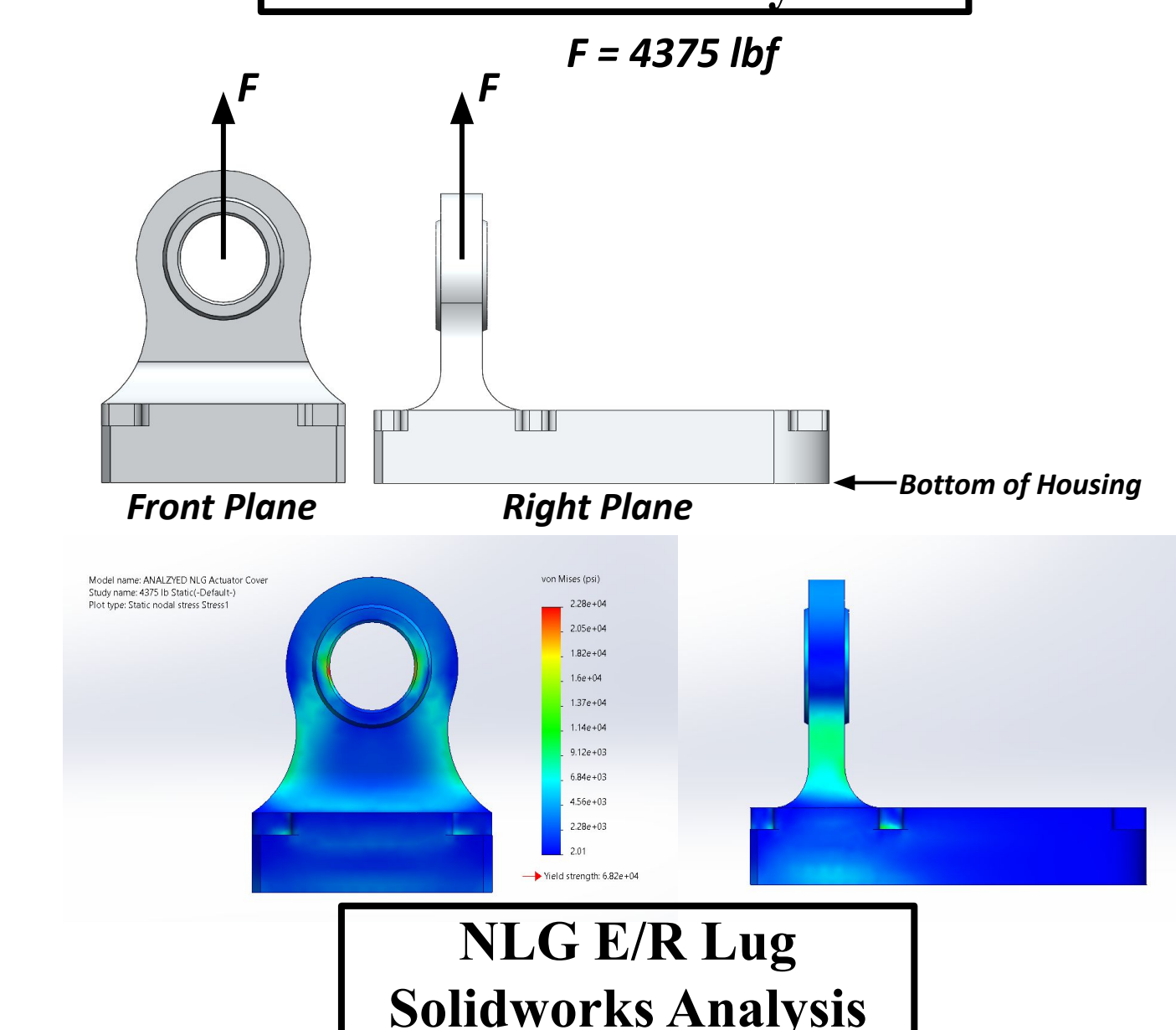
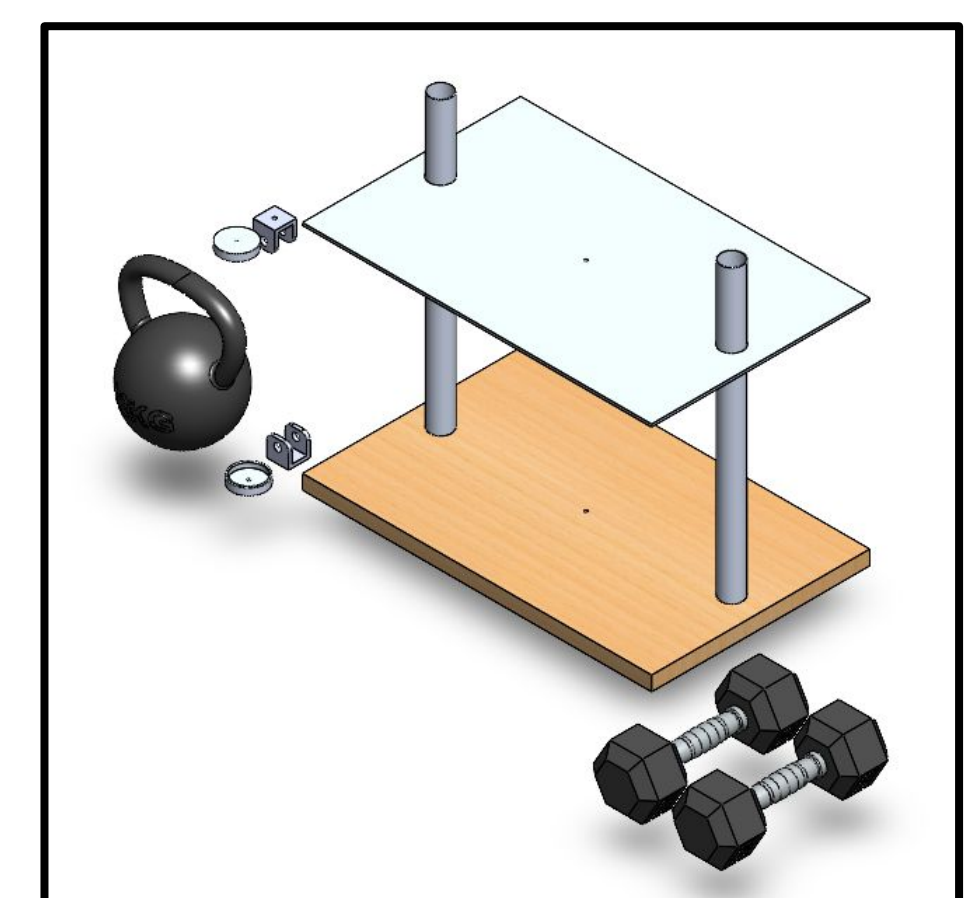
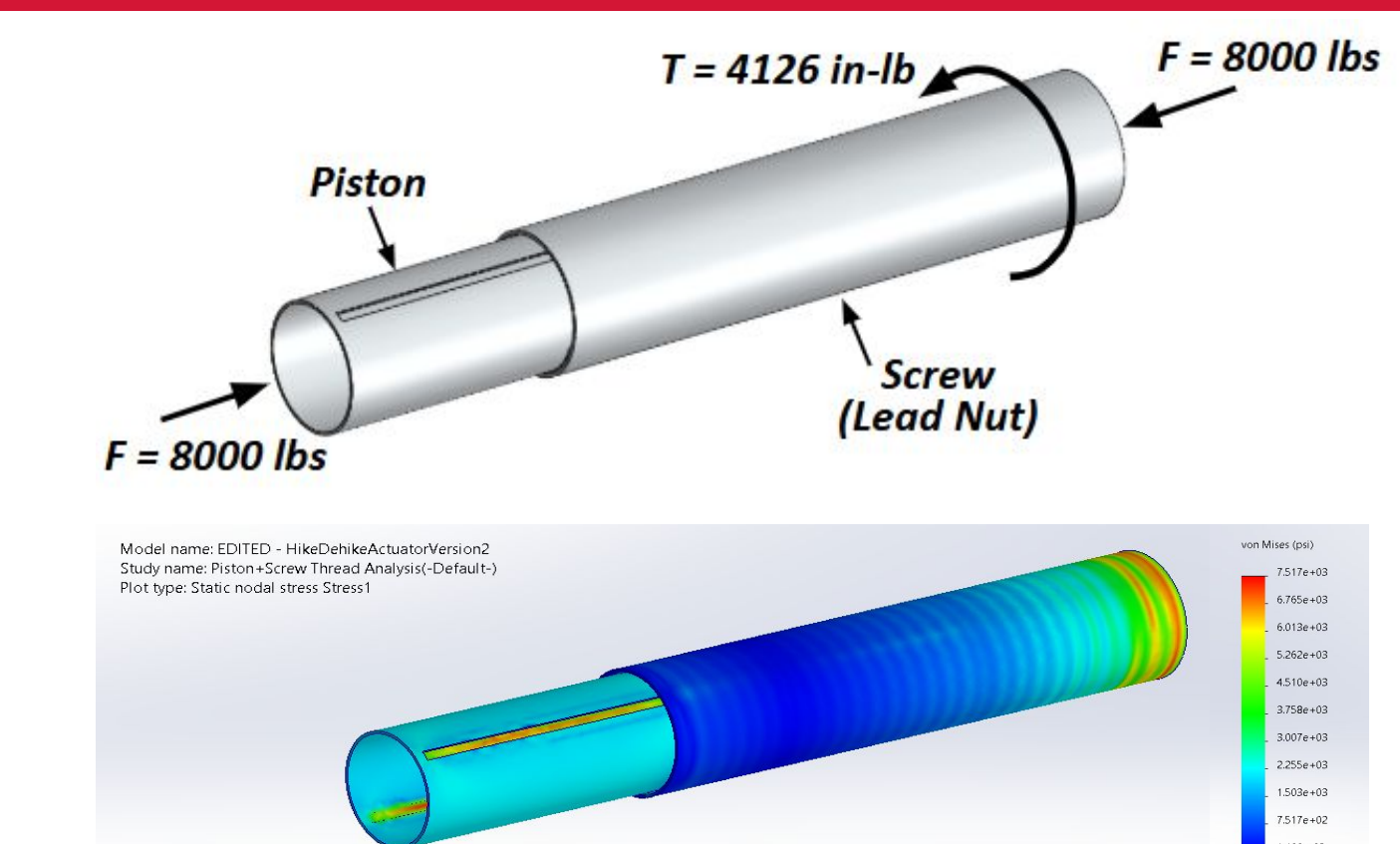
### Full-Scale CAD Assemblies



### Scale-Model Prototypes



### Testing & Analysis



### Acknowledgements

Team ACTUS would like to thank Dr. Shaffar from San Diego State University for arranging and advising this project. The team would also like to thank Andrew Simmons of Northrop Grumman for assistance as the project sponsor contact.

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